Thermal Study of HOM Couplers for Superconducting RF Cavities

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Abstract
The thermal behaviour of a "fish hook" type HOM coupler [1-2] for Superconducting RF Cavities is analyzed by numerical simulation using a 3D finite element based code for both CW (rings, recirculating linacs) and pulsed (s.c. colliders studies like TESLA) accelerator types. Consequently, a thorough study of the thermal stability of this system is necessary in both transient and steady-state regimes. Numerical simulations, assuming anomalous losses at the end of the HOM inner conductor, were performed. The effects of the HOM coupler geometry and the materials thermal conductivity on the critical heat flux inducing the quench of the system are discussed.

I. INTRODUCTION
The superconducting Higher Order Modes (HOM) couplers used for Superconducting Radiofrequency Cavities (SRF) cooled by Liquid Helium (LHe) are generally limited by thermal breakdown induced by various phenomena for accelerating fields $E_{acc}$ in the range 2-5 MeV/m when no efficient cooling of the inner conductor is provided. These $E_{acc}$ values are much smaller than the values to be reached for future projects (e.g. TESLA [3] : $E_{acc} = 25$ MeV/m). We consider that HOM and beam tubes are located inside the insulation vacuum of the LHe vessel; this indirect LHe cooling should obviously reduce the thermal quench limit of the HOM couplers. It is then necessary to choose properly the HOM coupler geometry and its construction material in order to increase its capability to withstand anomalous RF losses.

At low temperature, the RF surface resistance, the thermal conductivity ($k$), the specific heat per unit volume ($C_v$), the heat transfer coefficient at LHe-metal interfaces are strongly dependent on temperature ($T$). The resulting heat equation with the pertinent boundary conditions is then nonlinear. Moreover, the system studied is not axisymmetrical, hence a 3D computer code is necessary. We used the CASTEM 2000 code [4] for this purpose: this Finite Element Method (FEM) based code can handle such problems in both transient and steady-state conditions for arbitrarily shaped computational domains with multiple regions (i.e. materials).

II. METHOD AND ANALYSIS
A. The boundary value problem
The temperature field $T(x,y,z,t)$, where $x,y,z$ and $t$ are spatial-coordinates and time respectively, within the computational domain $\Omega$ is governed by the well-known non-linear 3D Heat Diffusion equation with no heat source:

$$\text{div}(k(T) \nabla T) = C_v \frac{\partial T}{\partial t}$$

where the subscript " $i$ " refers to thermal properties within the region $\Omega_i$.

For the cases of our concern, the typical boundary conditions are: prescribed heat flux $q$, heat transfer to the LHe at $T_{bath}$ controlled by the Kapitza conductance $H_K$ ($T_{bath} < T = 2.176$ K). For this parameter, we have used the following expression [5]:

$$H_K = 0.017 T_{bath}^{3/2} \left(1 + \frac{3}{2} \left(\frac{\Delta T}{T_{bath}}\right) + \frac{(\Delta T)^2}{4 T_{bath}} \right)$$

where $H_K$ unit is W/cm²/K, $\Delta T = T - T_{bath}$ and $T$ is the LHe-cooled surface temperature.

Concerning the transient computations, the whole system studied was assumed to be initially at the LHe temperature.

B. Variational problem and approximation
Using the so-called Galerkin Method, the semidiscrete problems corresponding to the above boundary value problem are derived. Briefly, this method consists on subdividing $\Omega$ into a set of subvolumes or finite elements, and expanding the solution in terms of FEM using $N$-components global basis functions. Then, a variational method is applied to these functions leading to $N$ differential (algebraic) equations for time-dependent (steady-state) problems respectively. These equations constitute the semidiscrete variational problem, where the unknown are precisely the nodal values $T_j(t)$ of the temperature field within the $\Omega$-mesh.

C. Resolution Method
In the steady-state case, the above non-linear system is solved by an iterative method using linearisation technique and a relaxation factor for improving the convergence of the process. The steady-state solution is reached when the relative variations of temperature at all the nodes of the $\Omega$-mesh is less than $10^{-3}$.

For the transient case, the above equations, discretized in time are solved by a n-step method: the temperature field at a given time is calculated from temperature fields at the n preceeding times (we chose n=2).
III. APPLICATION TO THE HOM COUPLER

A. The model and thermal stability criteria

The real system studied (Fig. 1) consists in the beam tube of the cavity, the HOM coupler (diam.: 40 mm and Inner Conductor (IC) diam.: 8 mm), the Nb flange of cryostat tank and the LHe cooled part of the cavity. For numerical simulation, this system was modelled using a slightly modified geometry (Fig. 2) which includes all the parts described previously up to the first iris of the cavity. Notice however that the thermal contact resistances at the sealings (i.e. flanges of the HOM) were neglected, the fish-hook has been replaced by a straight Inner Conductor (IC) which has no influence on the numerical results because the thermal radiations are negligible (T < 10 K), and we have also neglected the heat flux coming from the RF coaxial cable (using a simple Nb cover in place of the upper part of the HOM).

In the defect free case, the total RF losses dissipated on the IC of the HOM remain lower than 0.4 mW (with a peak heat flux density of 40 μW/cm²) for T_bath = 2.0 K, \( E_{acc} = 25 \text{ MeV/m} \) and a residual surface resistance \( \rho_{res} = 20 \text{ nΩ} \) [6]. These very low RF dissipations can not explain an increase of the hot spot temperature beyond the Nb critical temperature \( T_c = 9.2 \text{ K} \). Consequently, anomalous RF dissipation sources (cavity electron loading, surface defect, multipactor, ...) have to be considered as possible causes of the HOM coupler thermal breakdown. We have then assumed a highly dissipative area, located at the end of the IC, the farthest point from the cold source. With this pessimistic assumption we will obtain the minimum heat flux limit \( Q_c \) before the quench.

The thermal stability will be insured if two conditions are fulfilled: the temperature \( T_{max} \) at the hot spot must be lower than \( T_c = 9.2 \text{ K} \) and the maximum heat flux density transferred to the LHe bath must be below the critical heat flux inducing the Film-Boiling limit.

B. Steady state regime

The steady-state temperature distribution was computed up to the critical heat flux \( (Q_c) \) for the following arrangements and the results are summarized in Table 1: # 1. the whole HOM coupler-cavity assembly is machined from bulk Niobium with \( RRR = 40, 194 \) or \( 570 \), # 2. all these parts are in bulk Niobium (RRR = 40,194 or 570) excepted the IC and the Stub which are made of sputtered Niobium onto a Copper substrate of \( RRR = 300 \), # 3. the same material as in item#1 but with addition of a copper \( (Cu: RRR=300) \) thermal shunt between the Coupler and the LHe cryostat flange (see Fig. 2), # 4 the same arrangement as in item #2 with addition of the thermal shunt described in item #3.

<table>
<thead>
<tr>
<th>Nb quality</th>
<th>Arrangement</th>
<th>Critical heat flux ( Q_c ) (mW)</th>
</tr>
</thead>
<tbody>
<tr>
<td>RRR 40</td>
<td>#1</td>
<td>72</td>
</tr>
<tr>
<td>RRR 194</td>
<td>#1</td>
<td>314</td>
</tr>
<tr>
<td>RRR 570</td>
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<td>925</td>
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<td>RRR 40</td>
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<td>600</td>
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<tr>
<td>RRR 194</td>
<td>#2</td>
<td>1860</td>
</tr>
<tr>
<td>RRR 570</td>
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<td>3418</td>
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<tr>
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<td>347</td>
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<td>#3</td>
<td>1017</td>
</tr>
<tr>
<td>RRR 40</td>
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<td>2700*</td>
</tr>
<tr>
<td>RRR 194</td>
<td>#4</td>
<td>3328*</td>
</tr>
<tr>
<td>RRR 570</td>
<td>#4</td>
<td>5430*</td>
</tr>
</tbody>
</table>

Table 1: Critical heat flux inducing the HOM coupler thermal breakdown in the steady-state regime.

* Quench limited by the transition to film boiling on the LHe-cooled Nb flange (heat flux density: 2.4 W/cm²).

Analysing the temperature profiles along the IC for the arrangements #1 and #3 with \( Q = Q_c \), we note that the non-linear effects increase with the Nb RRR, the temperature difference along the IC is around 3.3 K whatever the RRR may be, there is no sensible effect of the thermal shunt on \( Q_c \) which increases (from nearly 70 mW to 1 W) with the Nb RRR (see Table 1). The values of the

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thermal impedance of the IC (Rth = ΔTIC/Qc) are 46 K/W, 10.5 K/W and 3.6 K/W for RRR = 40, 194 and 570 respectively. So, even in the best arrangement (#3 and Nb RRR = 570), the system is limited by the Niobium thermal conductivity.

In order to overcome this limitation, the arrangement #2 was examined, leading to a reduction by a factor 6 of the IC thermal impedance. The adding of a thermal shunt (arrangement #4) improves greatly Qc (Fig. 3) due to the sufficiently small thermal impedance (0.6 K/W) of the IC, and in this computation, 70% of Qc is derived through the thermal shunt.

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C. Transient regime

This study was limited to the arrangement #1 with RRR = 194. The applied heat flux was pulsed close to the TESLA cycle (pulse length tP = 2 ms : frep = 10 Hz). The maximum hot spot temperature during the RF pulse increases up to 8.6K at the end of the first RF pulse for Qc=6W. does not recover its initial temperature (i.e. Tbath=1.8K) at the end of the first pulse (4.4K), but reaches its stationary value (9.2K) in nearly 10 pulses (Fig. 4). The higher temperature after an infinity of pulses as function of the heat flux can be evaluated with:

\[ T_{\text{max}} \approx 3.63 Q^{0.35} + 0.31 Q^{0.49} \]

IV. CONCLUSION

A HOM coupler heat load capability of over 5 W has been reached in case of a continuous anomalous RF dissipation, by using an IC and a stub made with sputtered Nb onto a Cu substrate, and adding a thermal shunt connected to the LHe flange. In the transient regime, the maximum heat load sustained by the bulk Nb coupler is 6 W. In all the cases and for the corresponding critical heat fluxes, the maximum temperature at the iris of the cavity does not exceed 0.1 K with respect to Tbath, so that the thermal stability of the cavity is also insured.

We plan some thermal measurements on a special cavity/HOM assembly, to confirm these calculations and to take into account, if necessary, the effect of the contact resistances at the sealings.

V. REFERENCES